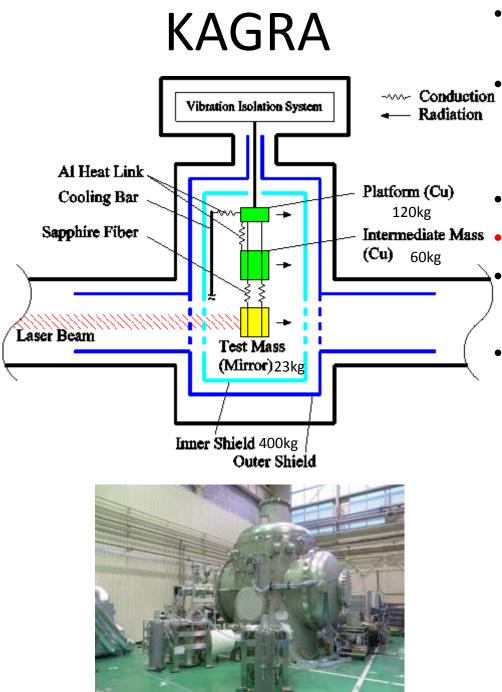


#### Verification of cooling time reduction for KAGRA

#### ICRR, Univ. of Tokyo, KEK<sup>A</sup>

Yusuke SAKAKIBARA N. Kimura<sup>A</sup>, T. Suzuki<sup>A</sup>, K. Yamamoto, D. Chen, S. Koike<sup>A</sup>, C. Tokoku, T. Uchiyama, and K. Kuroda

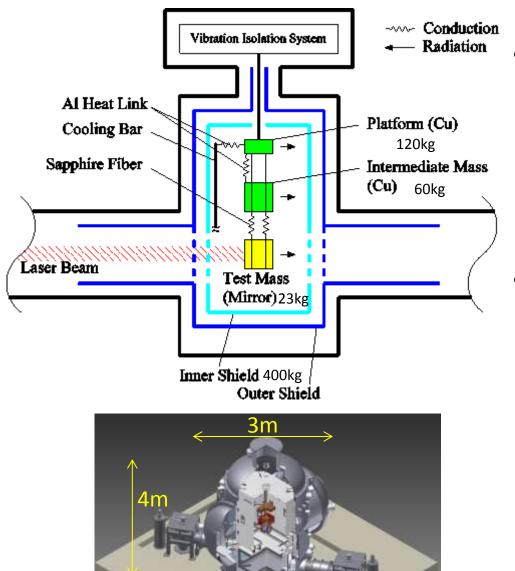
2013.4.19 Bilateral Italy-Japan Workshop (Italian Institute of Culture)



- Test masses is surrounded by double radiation shields
- Four cryocoolers per one mirror
  - Two cool inner shield (~400 kg)
  - Two cool bar to masses (~300 kg)
- It takes 2 months to cool down mirror
  - It is necessary to reduce cooling time
  - Cf. heating up by nitrogen (~10 days)
- How to reduce cooling time
  - Gas cooling
    - » Difficulty: mechanism to close and open holes of shield
  - Heat switch
    - » Difficulty: mechanism to contact and detach payloads
  - Increase radiation
    - » Slower but safer
    - » Adopted by KAGRA <sup>2</sup>

### Contents

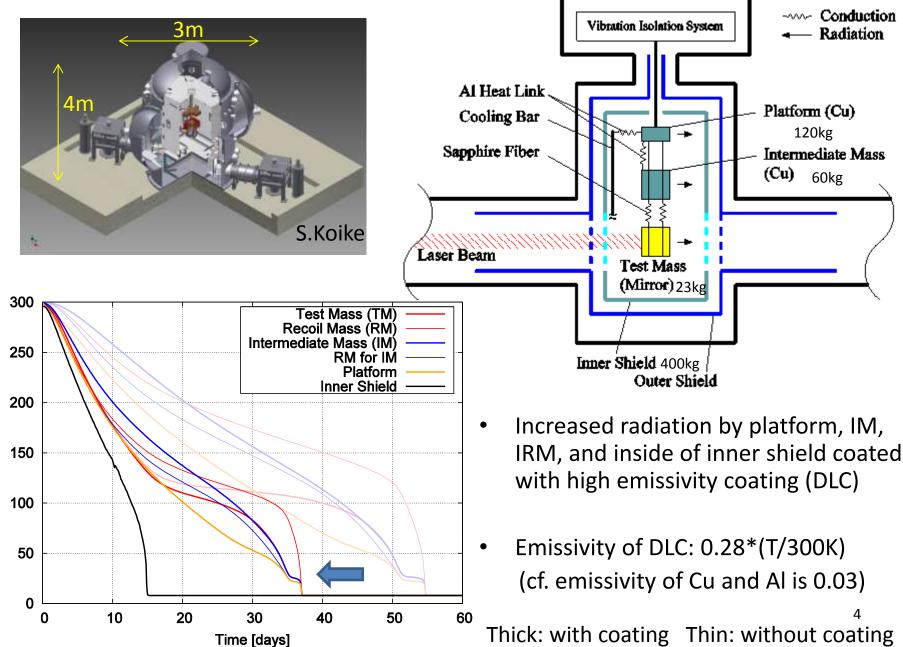
S.Koike



1

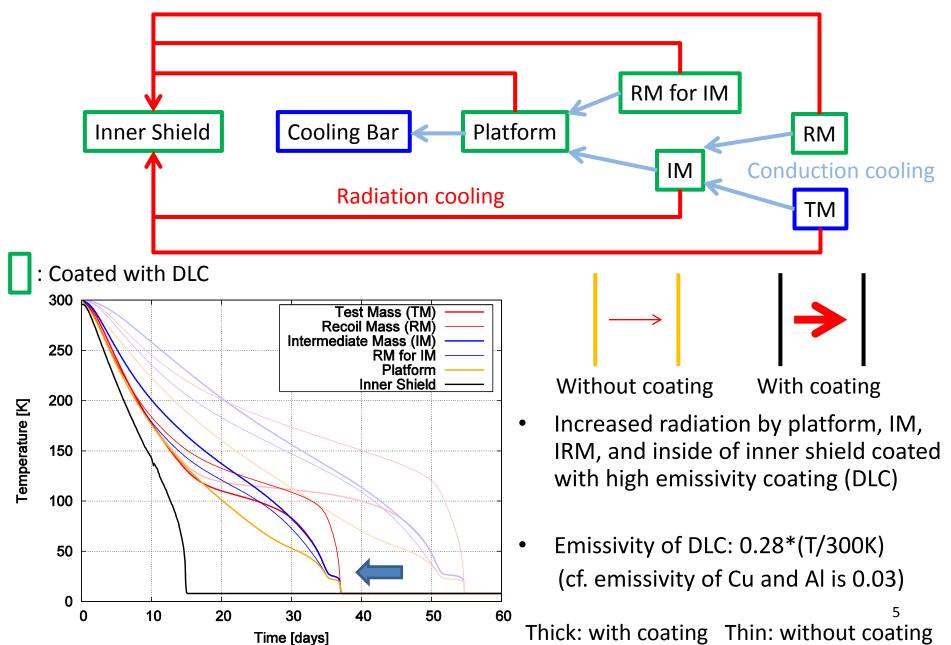
- Calculation of KAGRA cooling time
  - High emissivity coating
    (Diamond Like Carbon (DLC)
    coating) can reduce it to 40 days
- Experiment to verify
  - Calculation model of radiation
    - » Reflecting diffusely
    - » Reflecting specularly
  - Effect of high emissivity coating
  - Small experiment
  - Experiments in KAGRA cryostats

## Cooling time calculation of KAGRA



Temperature [K]

## Cooling time calculation of KAGRA



### Experiment

# Experiment with small apparatus

- Experimental test of calculation model of radiation and effect of high emissivity coating
  - Inner sphere (copper) is suspended inside outer sphere (aluminum)

Nylon wire

Pipe

Nitrogen dewar

Liquid nitrogen

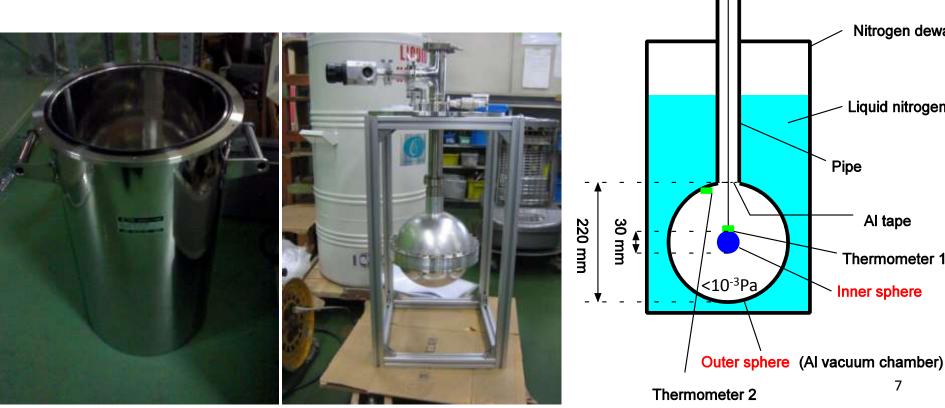
Al tape

**Inner sphere** 

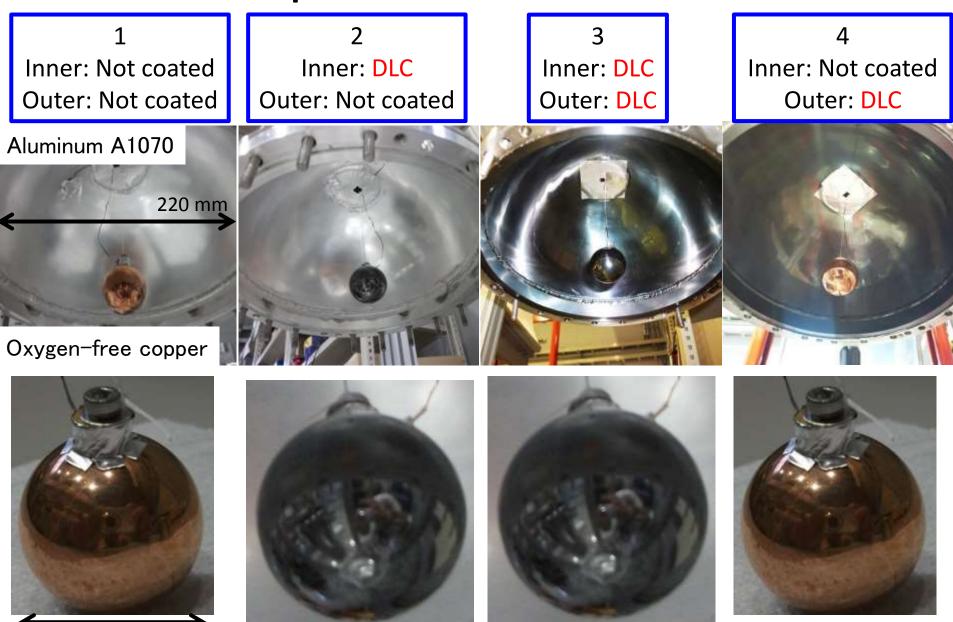
Thermometer 1

7

- Nylon wire: thermal conduction negligible
- Outer sphere is kept at 77 K using liquid nitrogen
- Cooling time is examined with and without coating



#### Samples to be measured

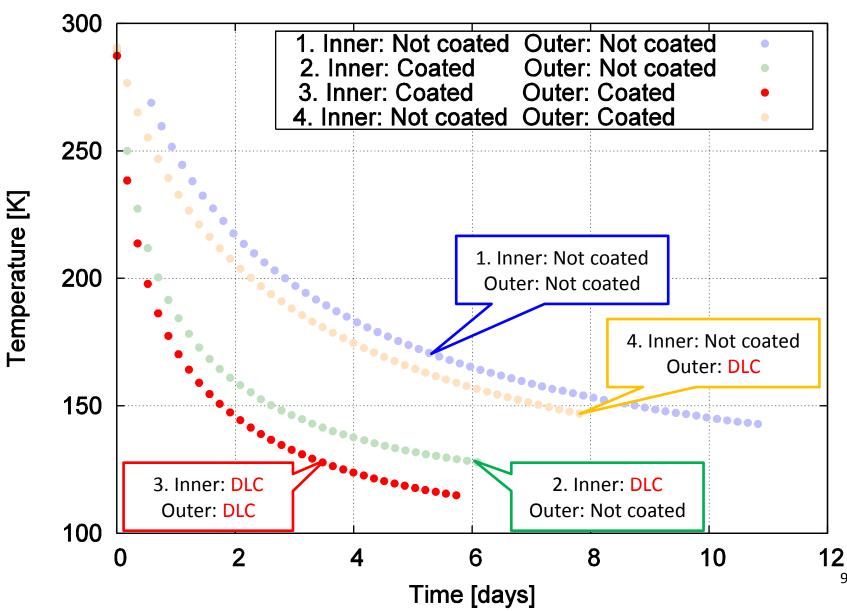


30 mm

8

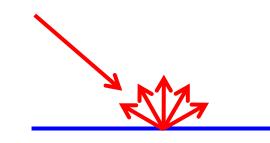
# Results (inner sphere: center)

• Fastest: both inner and outer spheres are coated

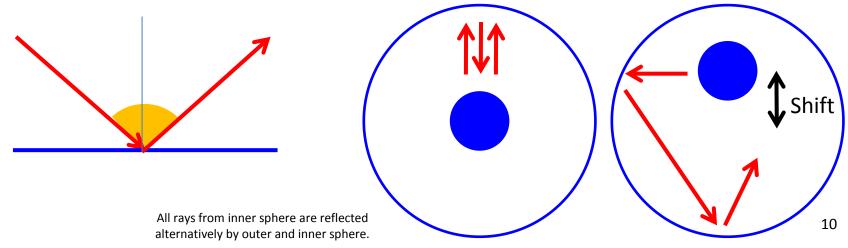


# Models of heat transfer via radiation

- Gray body model (diffusive reflection)
  - Surface reflects radiation to all angle
  - Heat transfer is independent of position of inner sphere

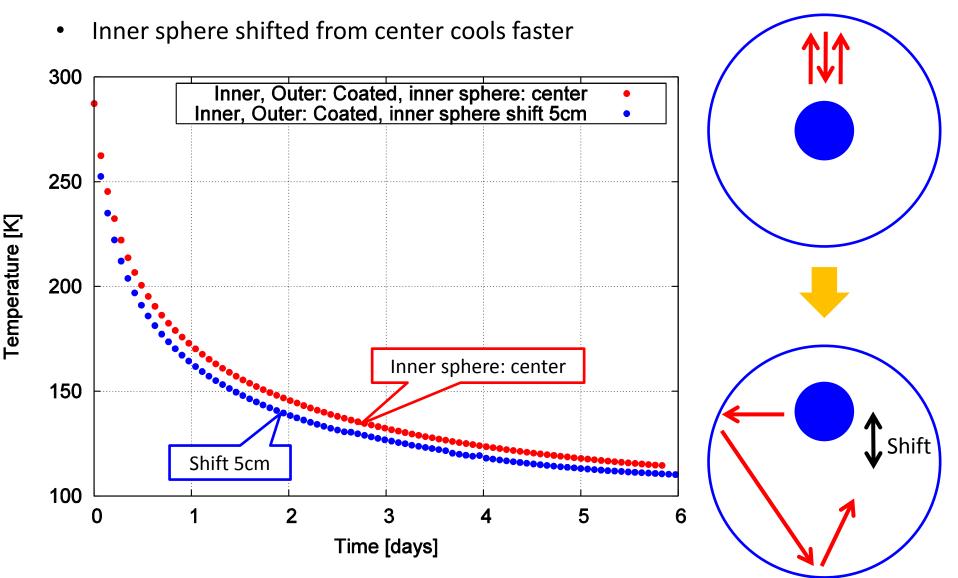


- Regular reflection model
  - Surface reflects radiation to the same angle as incident angle
  - Heat transfer depends on position of inner sphere



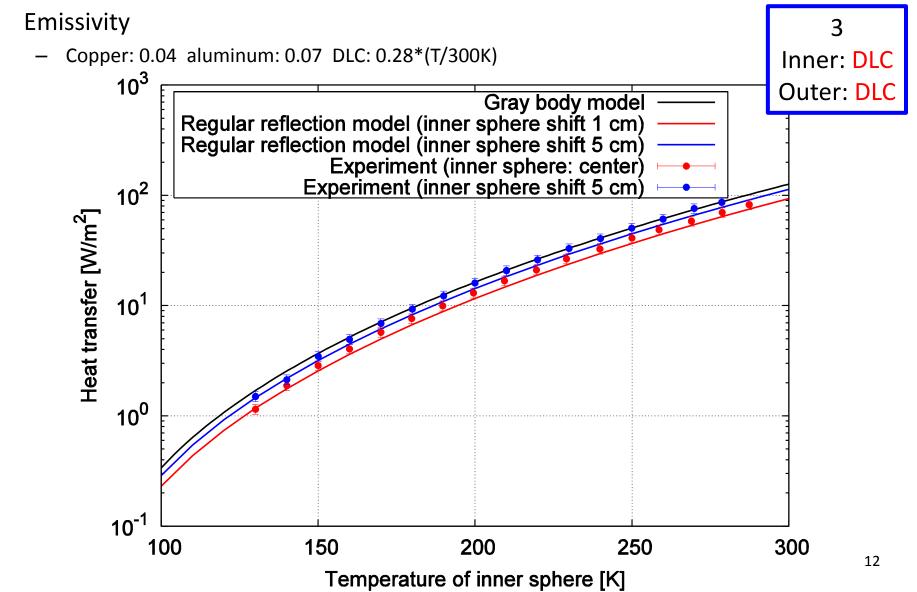
Some rays from inner sphere are reflected by outer sphere more than twice.

## Results (inner sphere: shifted from center)



#### Comparison with calculation models

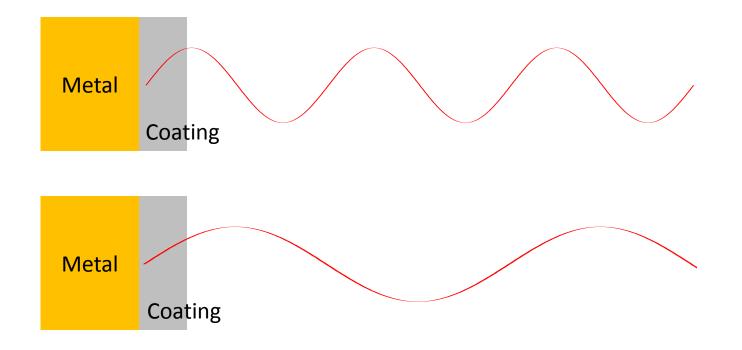
- Results are consistent with regular reflection model
  - Surface can be regarded as flat planes at wavelength of radiation (~10 um)



### Emissivity of coating (under discussion)

#### • Emissivity

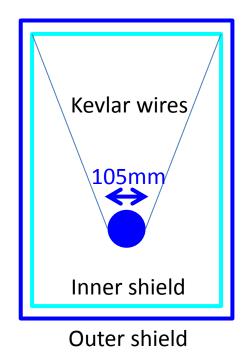
- Absorptivity at wavelength  $\lambda$  of radiation (10 um at 300 K,  $\lambda \propto T^{-1}$ )
- Assumption: constant absorption coefficient
  - Absorptivity  $\propto d/\lambda \propto T$ 
    - d: thickness of coating (1 um in this experiment)
  - Coating effectively looks thin for radiation with long wavelength



#### Experiments with KAGRA cryostats

# Experiments with KAGRA cryostats

- Cryostats made by Toshiba company
- Aluminum spheres are suspended inside KAGRA cryostats
  - Kevlar wires (aramide fibers): thermal conduction negligible
- Radiation was examined to verify scaling law

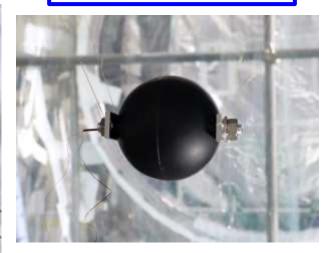




#### <u>Spheres in KAGRA cryostats</u>

Cryostat No.1 Aluminum sphere Not coated Cryostat No.2 Aluminum sphere Coated with DLC

Cryostat No.4 Aluminum sphere Coated with solblack



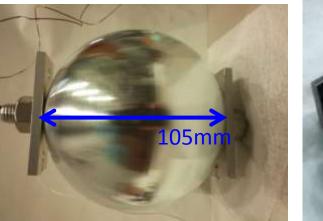


Spheres made by S.Koike, R.Kobayashi(KEK)



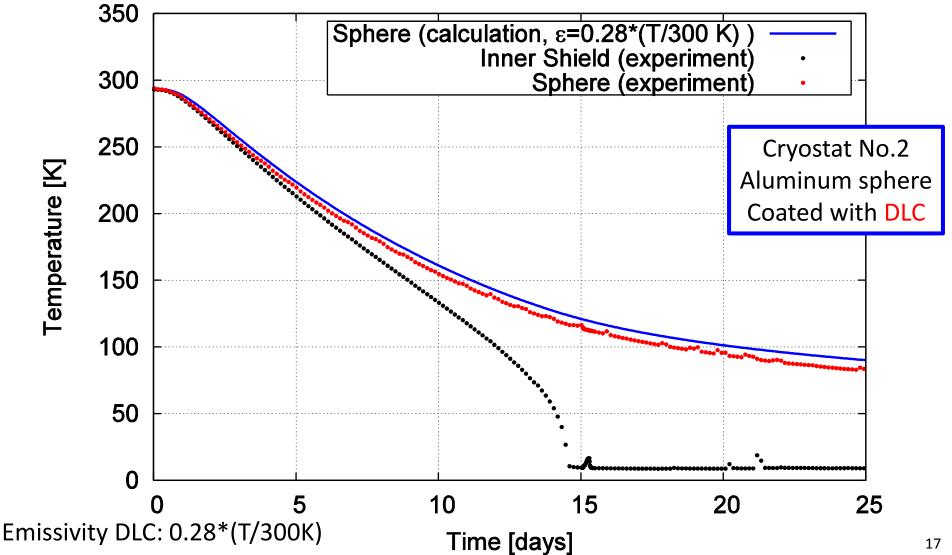
D. Chen





# Results

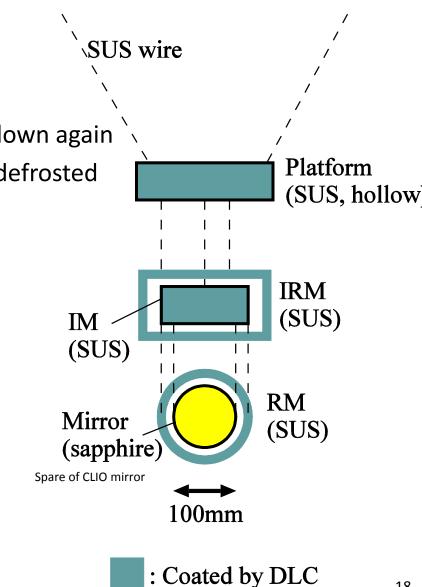
- Temperature of sphere calculated from temperature of inner shield
- Diffusive and regular reflection yield same results (not so symmetric shield)
- Results consistent with calculation



# Experiment with KAGRA cryostats

- 1/2 size dummy payload was suspended inside cryostat No.3
  - SUS wires: thermal conduction negligible
- Thermal radiation was examined
  - Cooling from room temperature
  - Payload heated up by heater and cooled down again
    - For cooling time when KAGRA mirror defrosted

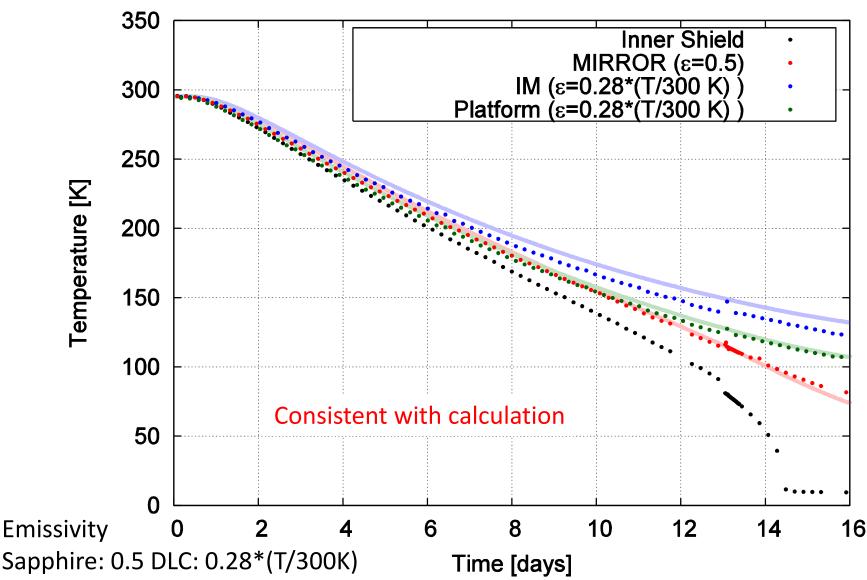




Payload designed and made by R.Kobayashi, S.Koike (KEK)

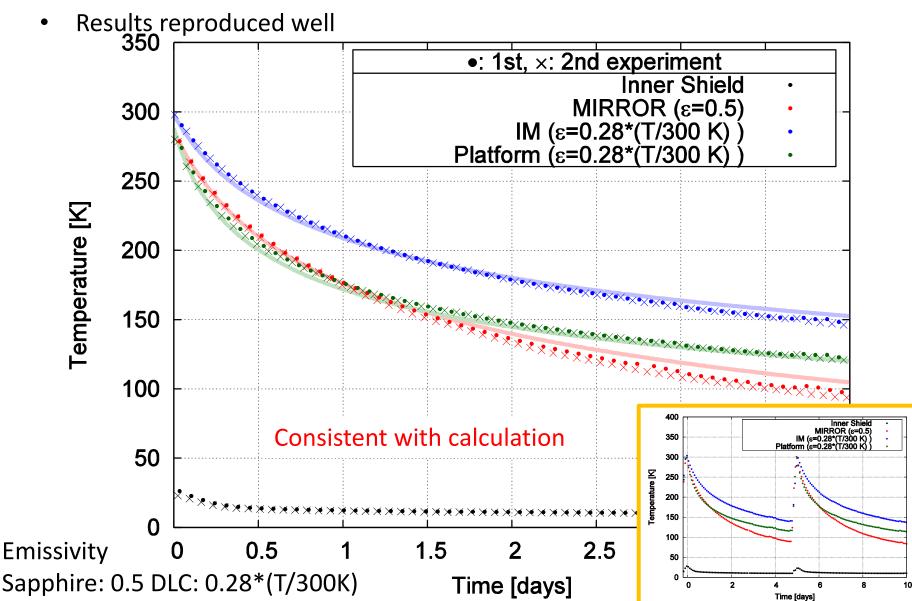
# Results (cooling from 300 K)

- Temperature of payloads calculated from temperature of inner shield
- Diffusive and regular reflection yield same results (not so symmetric shield)



# Results (heating up and cooling down) After shield cooled down completely

- Data for cooling time when KAGRA mirror defrosted



- Summary
  - Cooling time of KAGRA is calculated
  - Calculation model predicts cooing time correctly
    - Small experiment verified
      - Calculation model
      - Effect of high emissivity coating on reduction of cooling time
    - Experiment with KAGRA cryostats
      - Consistent with calculation (emissivity of sapphire: 0.5)
- Future work
  - R&D of another method to reduce cooling time